Energy Transport

A statement of the law of conservation of energy (for steady-state):



Example 8 Plane wall with heat source



Figure 1 Sketch illustrating one-dimensional conduction problem with heat generation

Energy balance around shell

Rate of energy in - Rate of energy out + Rate of energy production = 0

$$A \cdot q_x|_x - A \cdot q_x|_{x + \Delta x} + (A \Delta x)\dot{q} = 0$$

Divided through equation by $A\Delta x$ and taking limit $\Delta x \rightarrow 0$ we obtain

$$-\lim_{\Delta x \to 0} \frac{(q_x)|_{x+\Delta x} - (q_x)|_x}{\Delta x} = -\dot{q}$$

The expression on the left side is the first derivative of q_x with respect to x, so that equation becomes

$$\frac{d}{dx}(q_x) = \dot{q}$$

The equation may be integrated to give

$$\int d(q_x) = \int \dot{q} dx$$

$$q_x = \dot{q}x + C_1 \tag{1}$$

Apply Fourier's law:
$$q_x = -k \frac{dT}{dx}$$

 $-k \frac{dT}{dx} = \dot{q}x + C_1$
 $\int dT = \int \left(-\frac{\dot{q}}{k}x - \frac{C_1}{k}\right) dx$

The equation may be integrated to give

$$T = -\frac{\dot{q}}{2k}x^2 - \frac{C_1}{k}x + C_2$$
(2)

BC1: x = +L; $T = T_w$ BC2: x = -L; $T = T_w$ BC3: x = 0; $T = T_0$

Apply BC1 and BC2 into equation (2), respectively, we get

$$T_w = -\frac{\dot{q}}{2k}L^2 - \frac{C_1}{k}L + C_2$$
(3)

$$T_w = -\frac{\dot{q}}{2k}(-L)^2 - \frac{C_1}{k}(-L) + C_2 \tag{4}$$

Subtract equation (4) by equation (3) we obtain

$$C_1 = 0 \tag{5}$$

Substitute C_1 into equation (3) we get

$$C_2 = T_w + \frac{\dot{q}}{2k}L^2 \tag{6}$$

Substitute C_1 and C_2 into the equation (2) we obtain the temperature distribution as

$$T = -\frac{\dot{q}}{2k}x^{2} + T_{w} + \frac{\dot{q}}{2k}L^{2}$$

$$T = -\frac{\dot{q}}{2k}x^{2} + T_{w} + \frac{\dot{q}}{2k}L^{2}$$

$$T - T_{w} = -\frac{\dot{q}}{2k}x^{2} + \frac{\dot{q}}{2k}L^{2}$$
(7)

The temperature distribution is therefore

$$T - T_w = \frac{\dot{q}L^2}{2k} \left[1 - \left(\frac{x}{L}\right)^2 \right]$$
(8)

Alternative form of temperature distribution

BC3: x = 0; $T_0 = T_{max} \text{ or } \frac{dT}{dx} = 0$

Apply BC1 and then BC3 into equation (2) we get

$$C_1 = 0 \tag{9}$$

$$C_2 = T_0 \tag{10}$$

Substitute C_1 and C_2 into the equation (2) we obtain the temperature distribution as

$$T - T_0 = -\frac{\dot{q}}{2k}x^2$$
 (11)

Once the temperature and heat flux distributions are known, various information about the system may be obtained

(i) Maximum temperature rise (at x = 0)

$$T_{max} = T_0 \tag{12}$$

(ii) Average temperature rise

$$\overline{T} = \frac{T_{max}}{2} = \frac{T_0}{2} \tag{13}$$

(iii) Heat loss at the surface (for length L of plane)

$$Q_{loss} = 2Aq|_{x=L} = 2A(\dot{q}L) \tag{14}$$

Note: we got q from equation (1) and then substitute $C_1=0$